

Review for Exam 1

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- Drag Forces
- Cunningham Corrections
- Lift Forces
- Brownian Motion
- Diffusion Mechanisms
- Diffusion to a Cylinder

Drag Forces

Stokes



$$F = 3\pi\mu Ud$$

Drag
Coefficient



$$C_D = \frac{F_D}{\frac{1}{2}\rho U^2 A} = \frac{24}{Re}$$

Reynolds
Number



$$Re = \frac{\rho Ud}{\mu}$$

Drag Forces

$1 < Re < 1000$



$$C_D = \frac{24[1 + 0.15 Re^{0.687}]}{Re}$$

Cunningham Correction Clarkson University

For $1000 > Kn > 0$

Stokes-Cunningham Drag

$$F_D = \frac{3\pi\mu U d}{C_c}$$

Cunningham Correction

$$C_c = 1 + \frac{2\lambda}{d} [1.257 + 0.4e^{-1.1d/2\lambda}]$$

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Cunningham Correction Clarkson University

Variations of C_c with d for $\lambda = 0.07 \mu\text{m}$

Diameter, μm	C_c
10 μm	1.018
1 μm	1.176
0.1 μm	3.015
0.01 μm	23.775
0.001 μm	232.54

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Droplets Clarkson University

$$F_D = 3\pi\mu^f U d \frac{1 + 2\mu^f / 3\mu^p}{1 + \mu^f / \mu^p}$$

For Bubbles

$$F_D = 2\pi\mu^f U d$$

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Non-Spherical Particles Clarkson University

$$F_D = 3\pi\mu U d_e K$$

$$d_e = \left(\frac{6}{\pi} \text{Volume}\right)^{1/3}$$

K=Correction Factor

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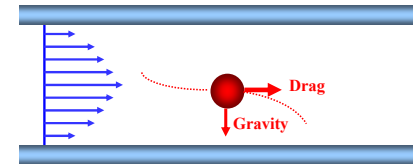
Correction Factor Clarkson University

Cluster Shape	Correction	Cluster Shape	Correction	Cluster Shape	Correction
oo	K = 1.12	oooo	K = 1.32	oo oo	K = 1.17
ooo	K = 1.27	ooooo	K = 1.45	o o o o o	K = 1.19
o o o	K = 1.16	oooooo	K = 1.57	oo oo oo	K = 1.17
oooooo o o	K = 1.64	ooooooo	K = 1.73		

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Aerosols Particle Motion Clarkson University



Equation of Motion

$$m \frac{du^p}{dt} = \frac{3\pi\mu d}{C_c} (u^f - u^p) + mg$$

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Aerosols Particle Motion Clarkson University

$$\tau \frac{du^p}{dt} = (u^f - u^p) + \tau g$$

Relaxation Time

$$\tau = \frac{mC_c}{3\pi\mu d} = \frac{d^2 \rho^p C_c}{18\mu} = \frac{Sd^2 C_c}{18\nu}$$

$$S = \frac{\rho^p}{\rho^f}$$

$$\tau(\text{s}) \approx 3 \times 10^{-6} d^2 (\mu\text{m})$$

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Terminal Velocity Clarkson University

$$u^p = (u^f + \tau g)(1 - e^{-t/\tau})$$

Terminal Velocity = Equilibrium Velocity after Large Time

$$u^t = \tau g = \frac{\rho^p d^2 g C_c}{18\mu}$$

$$u^t (\mu\text{m/s}) \approx 30 d^2 (\mu\text{m})$$

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Stopping Distance Clarkson University

Stopping Distance = Penetration distance for an initial velocity of u_0

$$\mathbf{u}^p = \mathbf{u}_0 e^{-t/\tau}$$

$$\mathbf{x}^p = \mathbf{u}_0^p \tau (1 - e^{-t/\tau})$$

$$\mathbf{x}^p = \mathbf{u}_0^p \tau$$

$$x^p (\mu\text{m}) \approx 3 d^2 (\mu\text{m})$$

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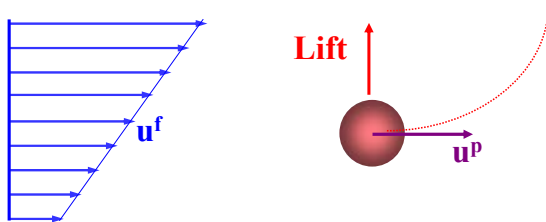
Particle Path Clarkson University

$$\mathbf{x}^p = \mathbf{x}_0^p + \mathbf{u}_0^p \tau (1 - e^{-t/\tau}) + (\mathbf{u}^f + \tau \mathbf{g}) [t - \tau (1 - e^{-t/\tau})]$$

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Lift Force Clarkson University



Saffman (1965, 1968)

$$F_{L(\text{Saff})} = 1.615 \rho \nu^{1/2} d^2 (u^f - u^p) \left| \frac{du^f}{dy} \right|^{1/2} \text{sgn} \left(\frac{du^f}{dy} \right)$$

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Saffman Lift Force Constraints Clarkson University

$$R_{es} = \frac{|u^f - u^p| d}{\nu} \ll 1$$

$$R_{e\Omega} = \frac{\Omega d^2}{\nu} \ll 1$$

$$R_{eG} = \frac{\dot{\gamma} d^2}{\nu} \ll 1$$

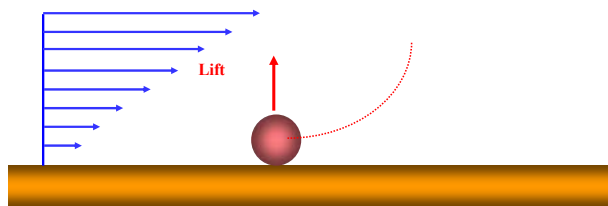
$$\varepsilon = \frac{R_{eG}^{1/2}}{R_{es}} \gg 1$$

McLaughlin (1991)

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Lift Force on a Particle Touching a Plane



Leighton and Acrivos (1985)

$$F_{L(L-A)} = 0.576\rho d^4 \dot{\gamma}^2$$

Saffman

$$F_{L(Saff)} = 0.807\rho v^{1/2} d^3 \dot{\gamma}^{3/2}$$

Lift Force in Turbulent Boundary Layer

Velocity Field in the Inertial Sublayer

$$u^+ = \frac{1}{\kappa} \ln y^+ + B$$

$$B \approx 5$$

$$30 < y^+ \leq 300$$

Wall Units

$$u^+ = \frac{u}{u^*}$$

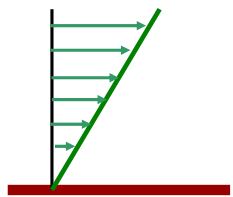
$$y^+ = \frac{u^* y}{\nu}$$

Viscous Sublayer

Turbulent stress is negligible

$$\tau_0 = \mu \frac{dU}{dy}$$

$$u^{*2} = \nu \frac{dU}{dy}$$



$$\frac{dU^+}{dy^+} = 1$$

$$u^+ = y^+$$

$$0 < y^+ \leq 5$$

Brownian Motion

Langevin Equation

$$\frac{du}{dt} + \beta u = n(t)$$

$$\beta = 3\pi\mu d / C_c m = 1/\tau$$

N(t) = White Noise

Spectral Intensity

$$S_{nn} = \frac{2kT\beta}{\pi m}$$

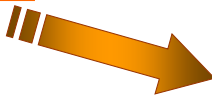
Brownian Motion Clarkson University

Mass Diffusivity



$$D = \frac{1}{2} \frac{d \overline{x^2}(t)}{dt}$$

Diffusivity



$$D = \frac{kT}{\beta m} = \frac{kTC_c}{3\pi\mu d}$$

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Computer Simulation Procedure Clarkson University

- $G_1 = \sqrt{-2 \ln U_1} \cos 2\pi U_2$
- $G_2 = \sqrt{-2 \ln U_1} \sin 2\pi U_2$
- Amplitude of the Brownian force is given by

$$n(t_i) = G_i \sqrt{\frac{\pi S_{nn}}{\Delta t}}$$

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Diffusion and Fick's Law Clarkson University

Fick's Law



$$J = -D \frac{dc}{dx}$$

Diffusion Equation

$$\frac{\partial c}{\partial t} + \mathbf{v} \cdot \nabla c = D \nabla^2 c$$

Diffusivity



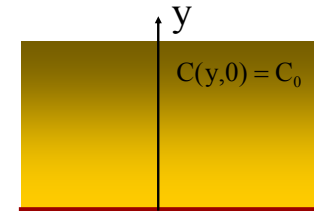
$$D = \frac{kTC_c}{3\pi\mu d}$$

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Particle Diffusion to a Wall Clarkson University

$$\frac{\partial c}{\partial t} = D \frac{\partial^2 c}{\partial y^2}$$



Similarity Variable

$$\eta = \frac{y}{\sqrt{4Dt}}$$

$$\frac{\partial c}{\partial y} = \frac{\partial c}{\partial \eta} \frac{\partial \eta}{\partial y} = \frac{\partial c}{\partial \eta} \frac{1}{\sqrt{4Dt}}$$

$$\frac{\partial^2 c}{\partial y^2} = \frac{\partial^2 c}{\partial \eta^2} \frac{1}{4Dt}$$

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Particle Diffusion to a Wall Clarkson University

$$\frac{\partial c}{\partial t} = \frac{\partial c}{\partial \eta} \frac{\partial \eta}{\partial t} = \frac{\partial c}{\partial \eta} \frac{-y}{2t\sqrt{4Dt}} = -\frac{\partial c}{\partial \eta} \frac{\eta}{2t}$$

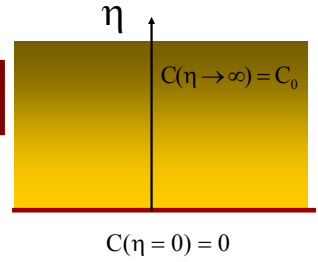
Similarity Equation

$$\frac{d^2 c}{d\eta^2} + 2\eta \frac{dc}{d\eta} = 0 \implies \ln\left(\frac{dc}{d\eta}\right) = -\eta^2 + \ln A$$

$$\frac{dc}{d\eta} = Ae^{-\eta^2} \implies c = A \int_0^\eta e^{-\eta_1^2} d\eta_1 + B$$

Particle Diffusion to a Wall Clarkson University

$$C(y, t) = C_0 \operatorname{erf}(y / \sqrt{4Dt})$$



$$\operatorname{erf}(\xi) = \frac{2}{\sqrt{\pi}} \int_0^\xi e^{-\xi^2} d\xi$$

$$\operatorname{erf}(0) = 0$$

$$\operatorname{erf}(\infty) = 1$$

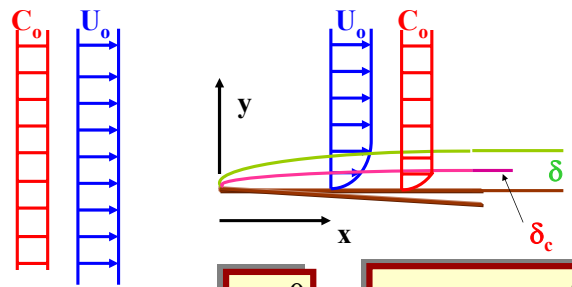
Particle Diffusion to a Wall Clarkson University

Diffusion Velocity $\implies u_D = \frac{J}{C_0} = \sqrt{\frac{D}{\pi t}} = \frac{D}{\delta_c}$

Diffusion Boundary Layer $\implies \delta_c = \sqrt{\pi Dt}$

Diffusion Force $\implies F_d = 3\pi\mu u_D / C_c$

Convective Diffusion to a Flat Plate Clarkson University



$$y = 0 \implies u = v = c = 0$$

$$y \rightarrow \infty \implies u = U_0, c = c_0$$

Convective Diffusion to a Flat Plate

Momentum

$$u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} = \nu \frac{\partial^2 u}{\partial y^2}$$

Mass

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

Concentration

$$u \frac{\partial c}{\partial x} + v \frac{\partial c}{\partial y} = D \frac{\partial^2 c}{\partial y^2}$$

Flat Plate - Similarity Variables

$$\eta = y \sqrt{\frac{U_0}{\nu x}}$$

$$\frac{u}{U_0} = f'(\eta)$$

$$\psi = \sqrt{\nu U_0 x} f(\eta)$$

$$c = c(\eta)$$

Momentum/Mass

$$ff'' + 2f''' = 0$$

Blasius Equation

Concentration

$$c'' + \frac{1}{2} S_c f c' = 0$$

$$S_c = \frac{\nu}{D}$$

Flat Plate - Similarity Variables

Boundary Conditions

$$f(0) = f'(0) = 0$$

$$f'(\infty) = 1$$

$$c(0) = 0$$

$$c(\infty) = c_0$$

Blasius Solution

$$\delta = 5 \sqrt{\frac{\nu x}{U_0}}$$

$$f''(0) = \gamma = 0.332$$

Near the Plate

$$f \approx \frac{\gamma}{2} \eta^2 + \dots$$

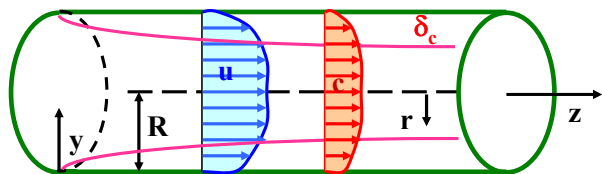
Concentration Profile

$$C = \frac{C_0 \int_0^\eta \exp(-\gamma_1 S_c z^3) dz}{\int_0^\infty \exp(-\gamma_1 S_c z^3) dz}$$

$$\gamma_1 = \frac{\gamma}{12}$$

$$\frac{c}{c_0} = \frac{\sqrt[3]{\gamma_1 S_c}}{0.89} \int_0^\eta [\exp(-\gamma_1 S_c z^3)] dz$$

Diffusion in a Tube Flow



Laminar Flow \Rightarrow $u = u_0 \left(1 - \frac{r^2}{R^2}\right)$ $y = R - r$

$u \sim u_0 \frac{2y}{R} + \dots$

Diffusion in a Tube Flow

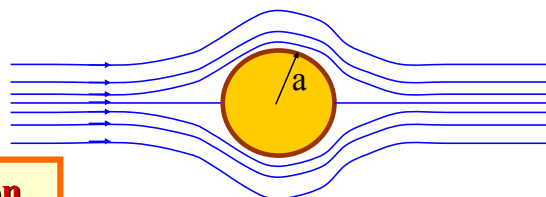
Diffusion Equation \Rightarrow $\frac{2u_0}{R} y \frac{\partial c}{\partial x} = D \frac{\partial^2 c}{\partial y^2}$

Boundary Condition
 $y=0, c=0$
 $y \rightarrow \infty, c=c_0$

Similarity Variable
 $\eta = \sqrt[3]{\frac{u_0}{DR} \frac{y}{x}}$

$c'' + \frac{2}{3} \eta^2 c' = 0$

Diffusion to a Cylinder



Diffusion Equation
 $\frac{v_\theta}{r} \frac{\partial c}{\partial \theta} + v_r \frac{\partial c}{\partial r} = D \left(\frac{\partial^2 c}{\partial r^2} + \frac{1}{r} \frac{\partial c}{\partial r} \right)$

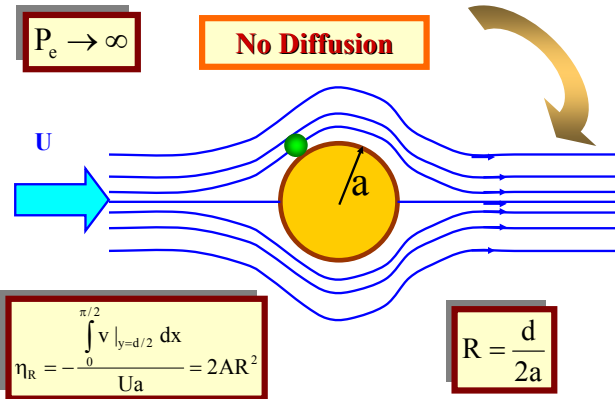
Boundary Conditions
 $r = a + \frac{d}{2}, c = 0$
 $r = \infty, c = c_\infty$

Diffusion to a Cylinder

Diffusion Equation
 $u \frac{\partial c}{\partial x} + v \frac{\partial c}{\partial y} = D \frac{\partial^2 c}{\partial y^2}$

Boundary Conditions
 $y=0, c=0$
 $y=\infty, c=c_\infty$

Direct Interception Limit Clarkson University



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Diffusion to a Cylinder Clarkson University

Let

$$\chi = \int \sin^{1/2} x_1 dx_1, \quad \psi_1 = \frac{\Psi}{2AaU}$$

Diffusion Equation

$$\frac{\partial c}{\partial \chi} = \frac{D}{aAU} \frac{\partial}{\partial \psi_1} \left(\psi_1^{1/2} \frac{\partial c}{\partial \psi_1} \right)$$

$$\psi_1 = 0, \quad c = 0$$

$$\psi_1 = \infty, \quad c = c_\infty$$

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Diffusion to a Cylinder Clarkson University

Similarity Equation

$$\xi = \frac{\psi_1}{\chi^{2/3}}$$

$$-\frac{AP_e}{3} \xi \frac{dc}{d\xi} = \frac{d}{d\xi} \left(\xi^{1/2} \frac{dc}{d\xi} \right)$$

$$c = \frac{c_\infty (AP_e)^{1/3}}{1.45} \int_0^{\sqrt{\xi}} \exp\left\{-\frac{2}{9} AP_e z^3\right\} dz$$

$$P_e = \frac{2Ua}{D} = R_c \cdot S_c$$

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Diffusion Clarkson University

- **Similarity Method**
- **Separation of Variable Method**
- **Integral Method**

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