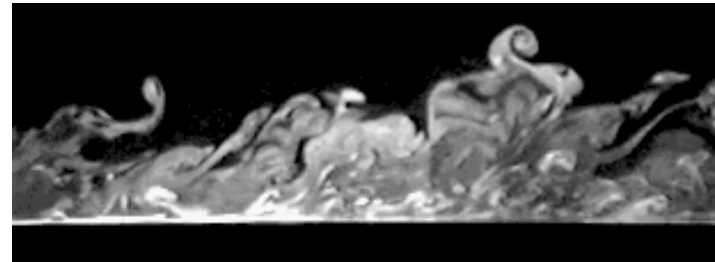


Turbulent Boundary Layer Flows

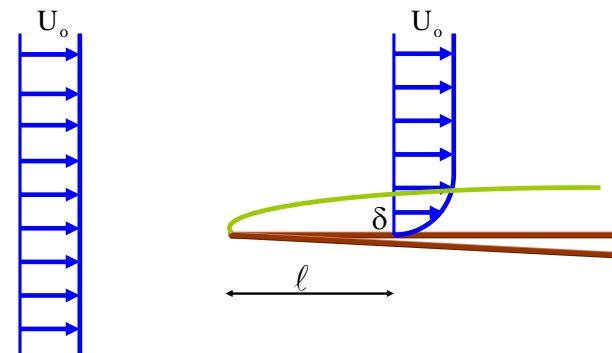
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Objectives

- Turbulent Boundary Layer Flows
- Friction Laws
- Power Laws
- Momentum Integral
- Blasius Resistance



Laminar Boundary Layer

Boundary Layer Thickness

$$\frac{\delta}{x} = 4.96 \text{Re}_x^{-\frac{1}{2}}$$

Friction Coefficient

$$C_f = 1.328 \text{Re}_\ell^{-\frac{1}{2}}$$

$$\text{Re}_\ell = \frac{U_0 \ell}{\nu}$$

Critical Reynolds Number

$$\text{Re}_{\text{crit}} \approx \begin{cases} 3.2 \times 10^5 \sim 10^6 \\ 5 \times 10^5 \sim 3 \times 10^6 \end{cases}$$

Turbulent Boundary Layer - Smooth

Boundary Layer Thickness

$$\frac{\delta}{x} = 0.37 \text{Re}_x^{-\frac{1}{5}}$$

Friction Coefficient

$$C'_f = 0.0577 \text{Re}_x^{-\frac{1}{5}}$$

$$5 \times 10^5 < \text{Re}_x < 10^7$$

Friction Coefficient including Transition

$$C_f = \frac{0.074}{\text{Re}_\ell^{\frac{1}{5}}} - \frac{A}{\text{Re}_\ell}$$

Turbulent Boundary Layer

Prandtl-Schlichting

$$C_f = \frac{0.455}{(\log \text{Re}_\ell)^{2.58}} - \frac{A}{\text{Re}_\ell}$$

$$\text{Re}_\ell > 10^7$$

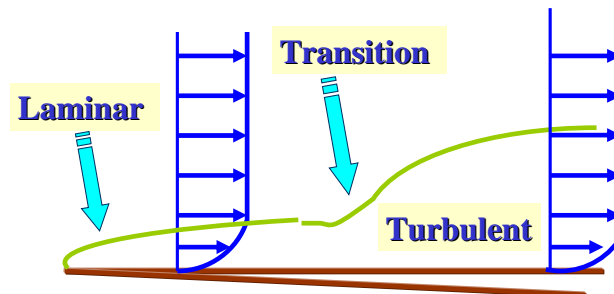
Schlichting Rough Zone

$$C_f = \left(1.89 + 1.58 \log \frac{\ell}{e} \right)^{-2.5}$$

Boundary Layer Thickness

$$\frac{\delta}{x} = 0.16 \text{Re}_x^{-\frac{1}{7}}$$

Turbulent Boundary Layer



Boundary Layer Transition Clarkson University

$$R_{\delta} |_{\text{critical}} \approx 1220$$

$$R_{\theta} |_{\text{critical}} \approx 420$$

$$R_L |_{\text{critical}} = 5 \times 10^5 (\sim 3 \times 10^6)$$

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Turbulent Boundary Layer Equations Clarkson University

Momentum Equation

$$U \frac{\partial U}{\partial x} + V \frac{\partial U}{\partial y} = -\frac{1}{\rho} \frac{dP_0}{dx} + \nu \frac{\partial^2 U}{\partial y^2} - \frac{\partial}{\partial y} \overline{u'v'}$$

Pressure

$$P_0 = \bar{P} + \rho \overline{v'^2}$$

Mass

$$\frac{\partial U}{\partial x} + \frac{\partial V}{\partial y} = 0$$

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Momentum Integral Method Clarkson University

$$\frac{d}{dx} \theta + \frac{2\theta + \delta^*}{U_0} \frac{dU_0}{dx} = \frac{\tau_0}{\rho U_0^2}$$

Displacement Thickness

$$\delta^* = \int_0^{\infty} \left(1 - \frac{U}{U_0}\right) dy$$

Momentum Thickness

$$\theta = \int_0^{\infty} \frac{U}{U_0} \left(1 - \frac{U}{U_0}\right) dy$$

Shape Factor

$$H = \frac{\delta^*}{\theta}$$

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Power Law Velocities Clarkson University

$$\frac{U}{U_0} = \left(\frac{y}{\delta}\right)^{\frac{1}{n}} = \eta^{\frac{1}{n}}$$

$$\frac{U_{av}}{U_0} = \frac{n}{n+1}$$

$$\frac{U_0}{u^*} = c_1 R_{\delta}^{\frac{1}{n+1}}$$

$$\frac{c_f}{2} = c_1^{-2} R_v^{-\frac{2}{n+1}}$$

$$\frac{\delta^*}{\delta} = \frac{1}{n+1}$$

$$\frac{\theta}{\delta} = \frac{n}{(n+1)(n+2)}$$

$$H = \frac{\delta^*}{\theta} = \frac{n+2}{n}$$

$$\frac{U_0 \theta}{\nu} = \left(c_2 \frac{U_0 x}{\nu}\right)^{\frac{n+1}{n+3}}$$

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Blasius Resistance Law Clarkson University

$$c_f \sim \left(\frac{U_0 \delta}{\nu} \right)^{-1/4} \quad \frac{U}{u^*} = 8.3 \left(\frac{u^* y}{\nu} \right)^{1/7}$$

$$\frac{U_0 - U}{u^*} = 0.6 \left(1 - \frac{y}{\delta} \right)^2 \quad \frac{y}{\delta} > 0.15$$

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Turbulent Boundary Layer Friction Coefficient Clarkson University

Ludwig-Tillmann →

$$C_f = 0.246 \times 10^{-0.678H} \left(\frac{U_0 \theta}{\nu} \right)^{-0.268}$$

$$10^3 < R_\theta < 10^4$$

$$H \approx 1.36$$

$$C_f \approx 0.03 R_\theta^{-0.268}$$

$$R_\theta = 0.045 R_x^{0.79}$$

$$n = 6.45$$

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Turbulent Boundary Layer Flows Clarkson University

Concluding Remarks

- Turbulent Boundary Layer Flows
- Friction Laws
- Power Laws
- Momentum Integral
- Blasius Resistance

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